



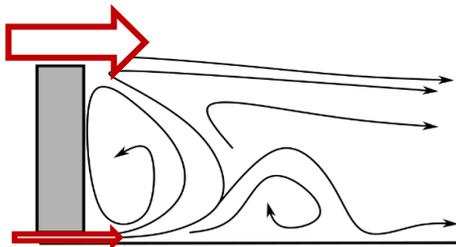
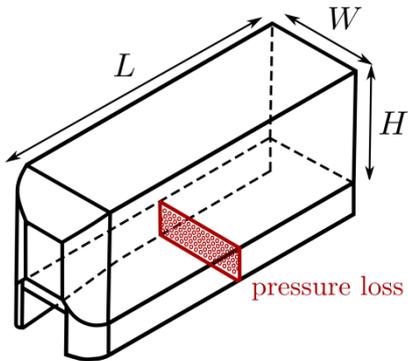
# FLOW CONTROL STRATEGIES FOR DRAG REDUCTION OF A MID-SCALE TRAILER MODEL

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# CONTROL STRATEGY ON A SIMPLIFIED TRAILER

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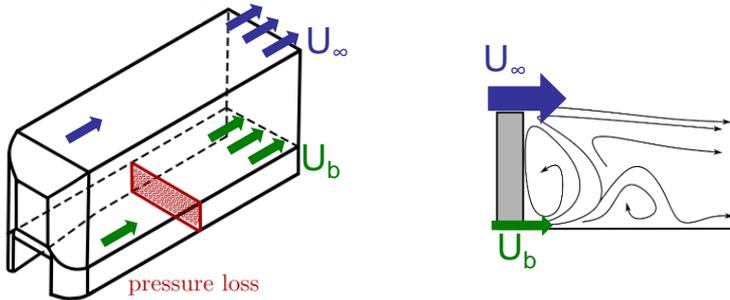
### Heavy truck's model



- Introduction:
  - Wake topologies related to the underbody momentum
  - Choice of one class for control
  
- Medium/high frequency forcing ( $St_{act} = 5-15$ )
  - Effects on mean field, deflection angle, pressure field
  - Changes within the turbulence
  
- Forcing strategy (**Local forcing**)
  - Influence on pressure recovery and drag reduction
  - How to improve the control efficiency?
  
- Conclusions & perspectives

# Wake topologies

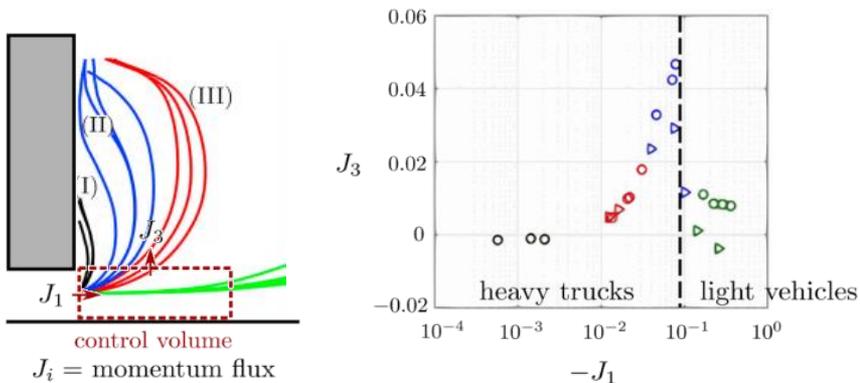
□  $G^* = \text{cte}$  then driving parameter  $\rightarrow$  ratio  $\lambda = U_b / U_\infty$



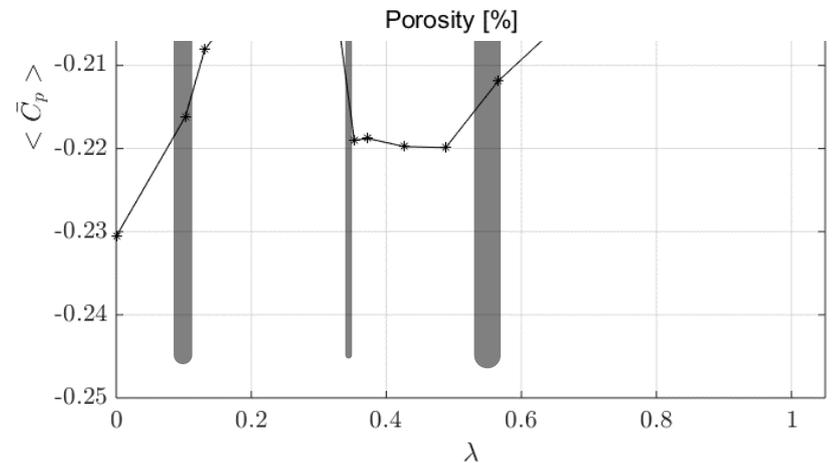
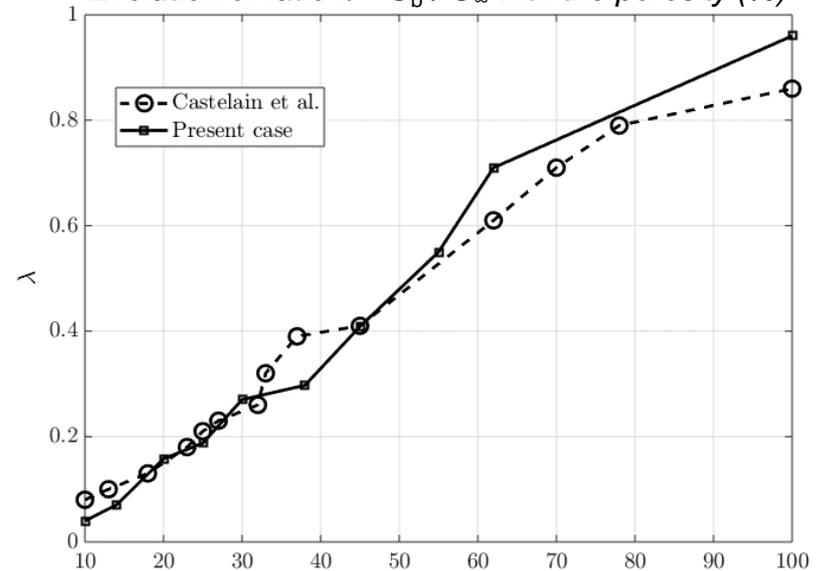
□ four wake topologies

Parameters for the topology classification into classes :

Momentum flux,  $C_p$ , gradients of pressure ...



Evolution of ratio  $\lambda = U_b / U_\infty$  with the porosity (%)



Castelain et al. JWEIA 2018

# Wake topologies: selection of one case of study

- ❑ One specific wake topologies
- ❑ **Present case ( $Re_H 4.10^5$ ) :**
  - **class 3 with  $\lambda \sim 0.4$**
  - $\lambda$  representative of real truck cases
  - unfavourable base pressure and drag

## Control investigations

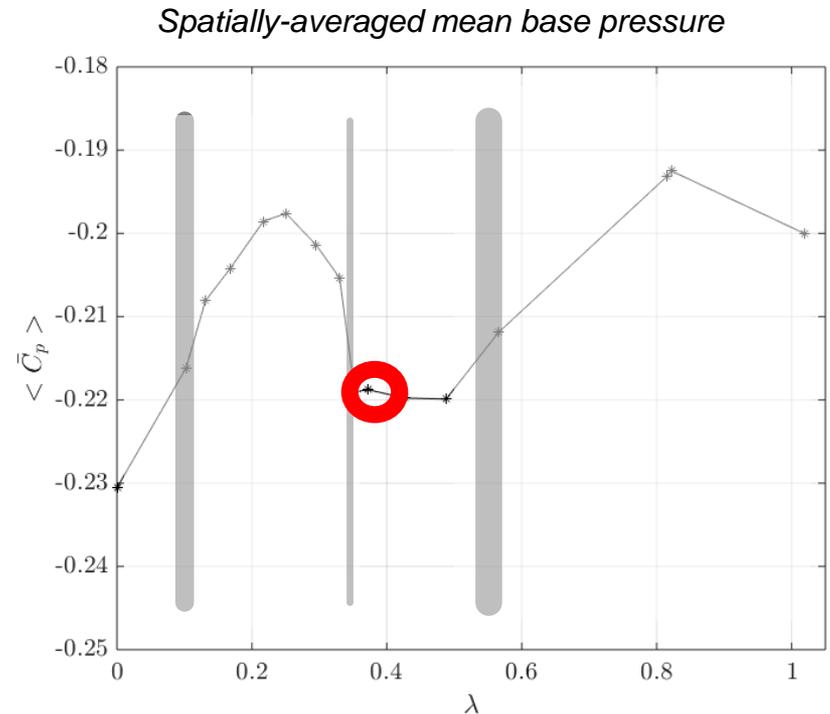
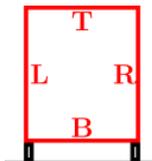
- ❑ *Previous at LMFA*  
(PhD work of Chaligné and Szmigiel)
  - control of flows : class 1 and 2

- ❑ flow modifications in present case :
  - open-loop global forcing
  - forcing parameters:

$$P_{rel} = 1,9 \text{ bar} \rightarrow C_{\mu} = DC N (sj/S) (V_{jmax}/U_{\infty})^2 = 3.10^{-2}$$
$$DC = 50 \%$$

$$f_{act} = 350 \text{ Hz} \rightarrow St_{act} = fH/U_{\infty} = 5$$

- **30%** of pressure recovery
- **10%** of drag reduction



# Medium and high frequency forcing : mean velocity

## **General features** of class 3 flows without control:

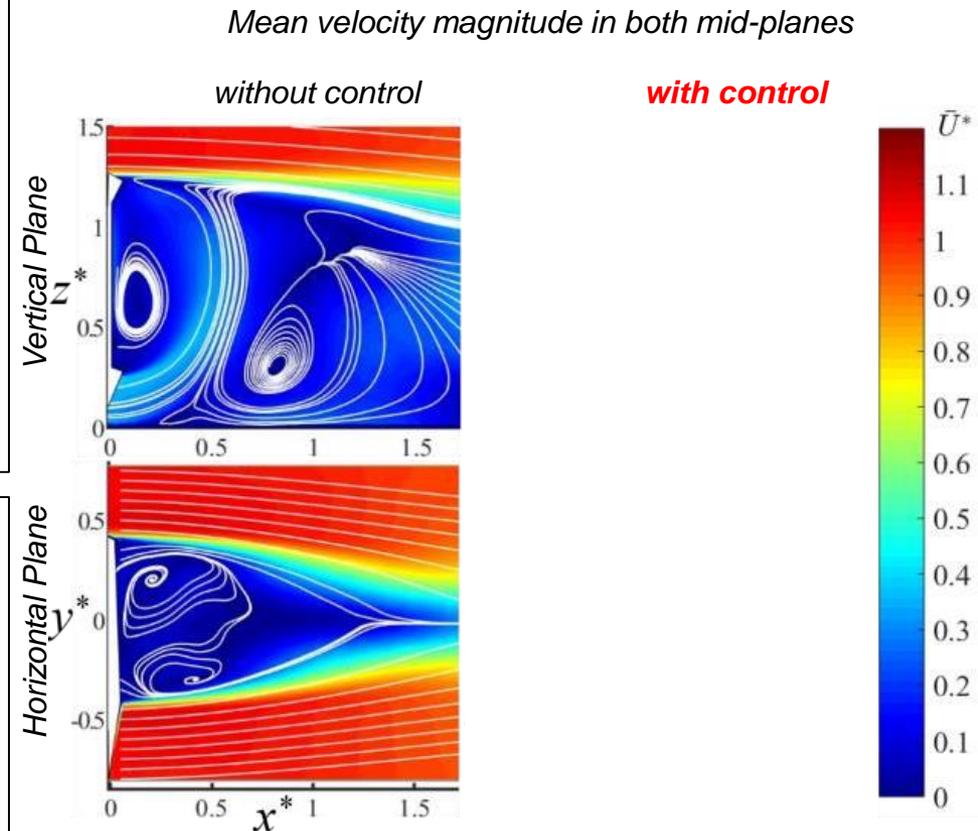
- mean curved jet separated from the ground:
  - fluid caught up near the base
  - secondary vortex
- back-flow due to underbody flow :
  - flux convected towards the upper shear layer
  - impinge and limit the growth of SL

## **With control**

- mean deflection of the potential flow on three sides of SL
- higher intensity on left and right side SL
- no significant effect of actuation in the curved jet

## **First conclusions**

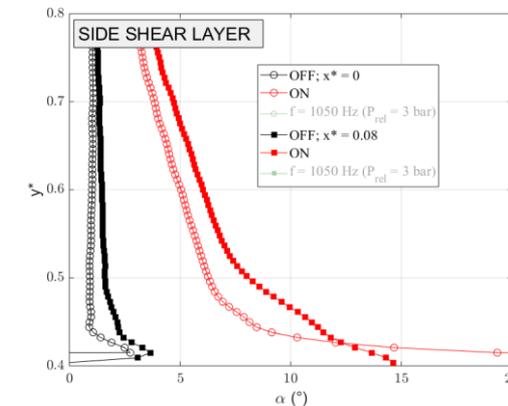
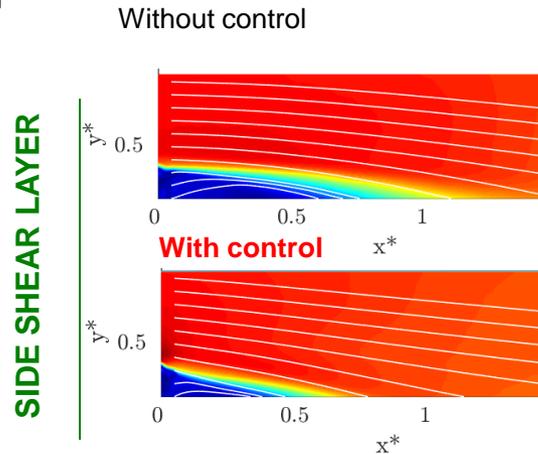
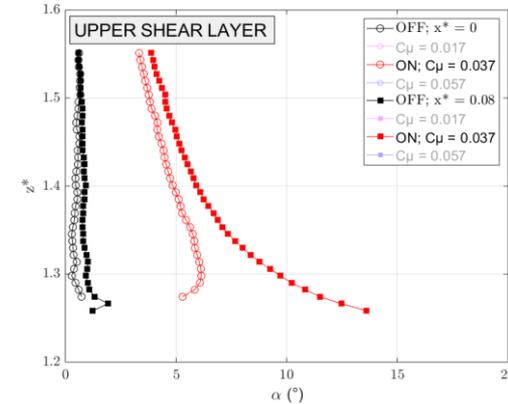
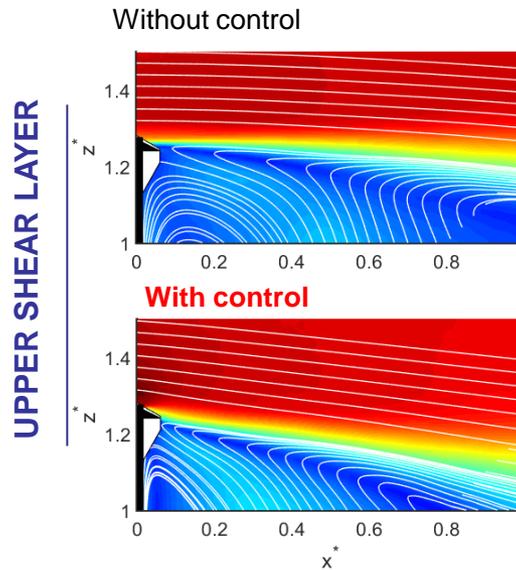
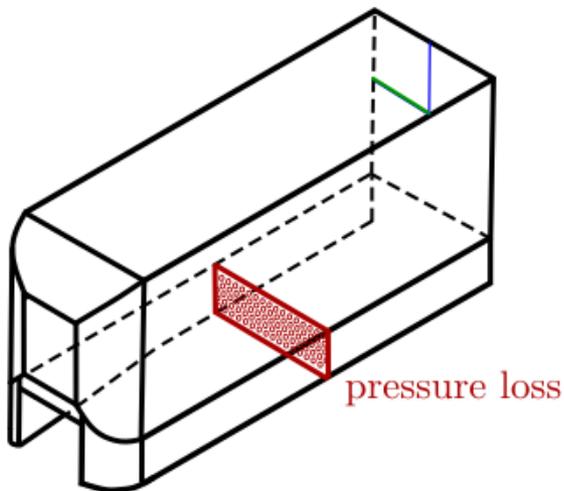
- **No flow class changes under actuation**
- **Vertical assymetry is only slightly modified (re-enforced)**
- Similar conclusions for a higher frequency forcing ( $f_{act} = 1050 \text{ Hz}$ ,  $St_{act} = 15$ )
- **Deflection of the potential flow is the main effect of actuation?**



# Medium and high frequency forcing : deflection of the potential flow

*with control*

- ❑ deflection at ( $x^* \sim 0$ ) :  $\alpha_{\text{horizontal}} > \alpha_{\text{vertical}}$
- ❑ global deflection of the wake
- ❑ analysis of streamlines ( $x^* = 0$  and  $0.08$ ) :
  - high flow deviation around flap location
  - max. deflection angle  $\sim$  flap angle
- ❑ similar results for  $St_{act} = 15$



# Medium and high frequency forcing : mean pressure

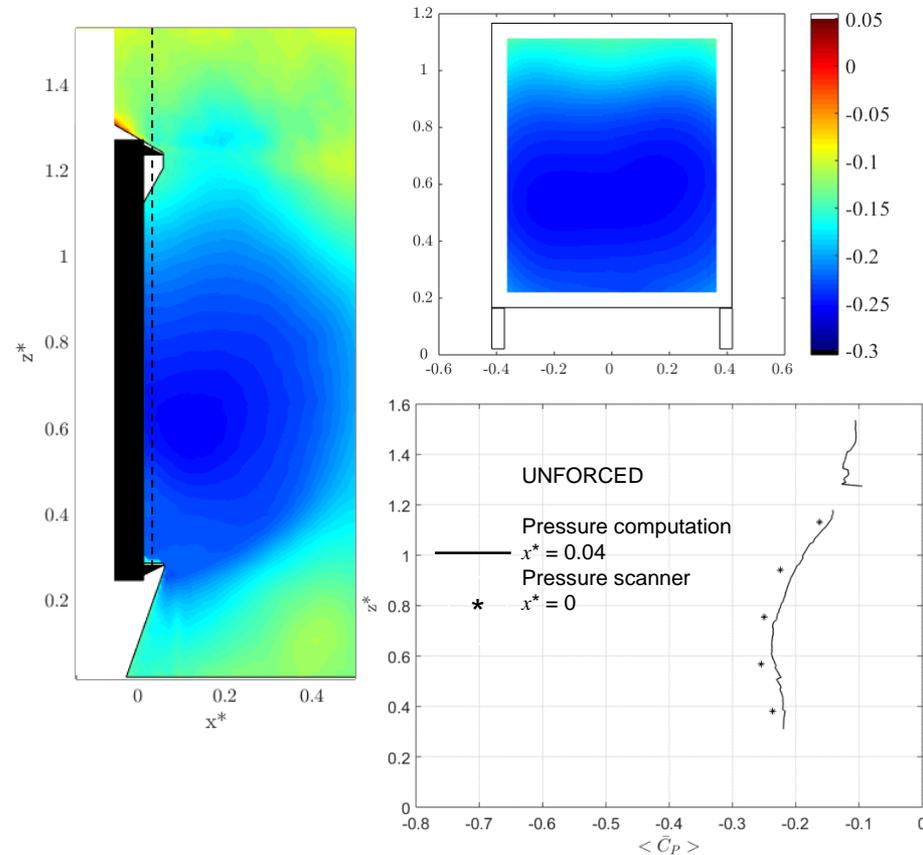
## Mean pressure 2D field

- 2D-2C PIV measurements
- 2D N-S averaged equations (incompressible)
- stochastic integration scheme
-  *Oxlade JFM 2015*
- reference pressure: Bernoulli equation along a streamline within the potential flow

without control :

- agreement with mean base pressure measurements
- base pressure vertically stratified
- large area with negative pressure trapped by the curved jet

Mean pressure map in the vertical mid-plane & in the base



# Medium and high frequency forcing : mean pressure

without control :

- base pressure vertically stratified
- large area with negative pressure

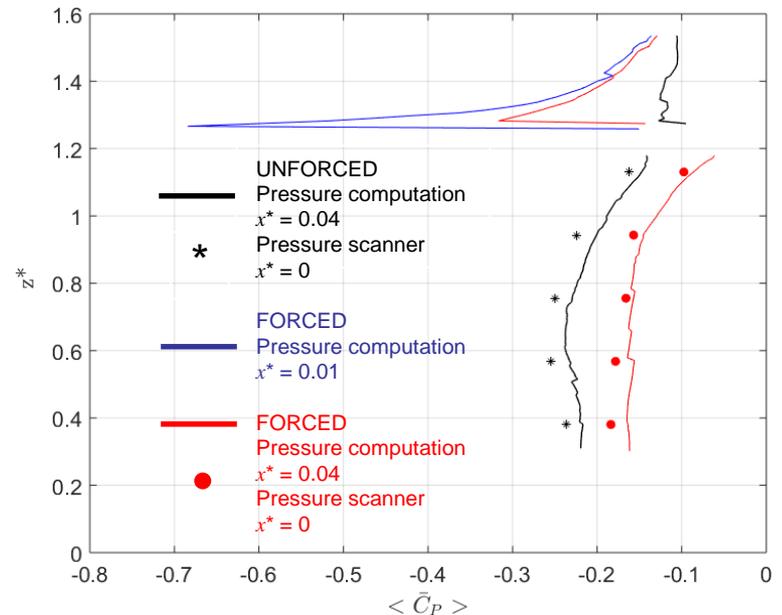
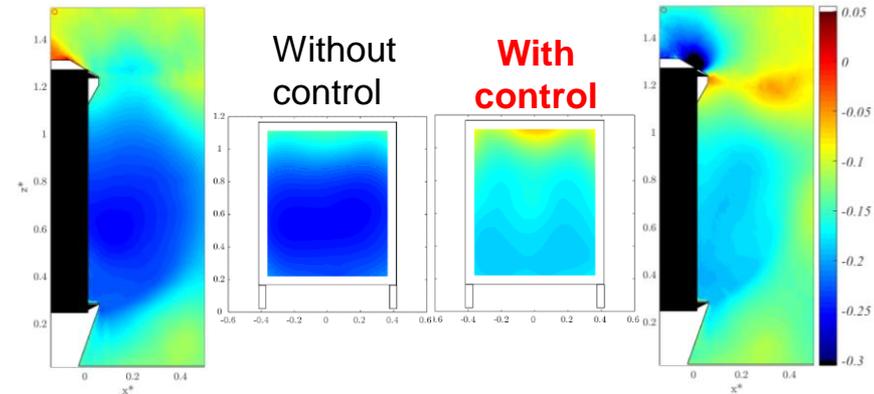
with control

- global pressure recovery in the near wake linked with the global shift of wall pressure
- local large negative pressure peak over the base edge and the flap
- faster pressure recovery of the potential flow in  $x$  direction

To be highlighted from this pressure analysis :

- good agreement with pressure measurements
- **large pressure peak due to combination of passive and active actuation**
- **important base pressure recovery ( $\gamma_p \sim 30\%$ )**

Mean pressure map in the vertical mid-plane & in the base



# Medium and high frequency forcing : effects on the curved jet

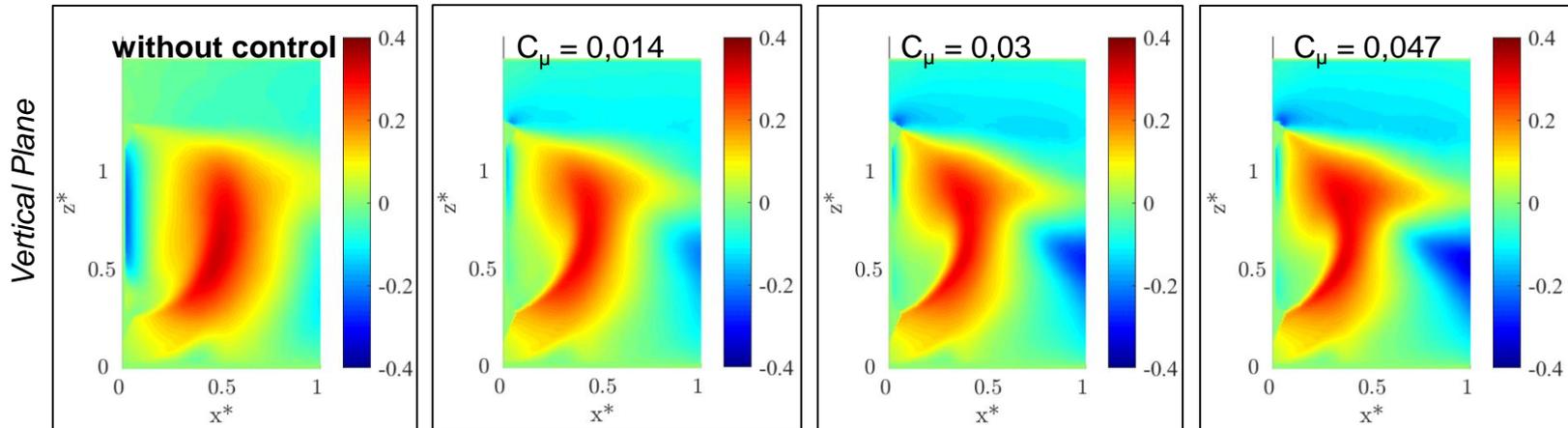
*without control*

- mass flux convected from the underbody flow towards the upper part of the base

*with control*

- increase of control intensity ( $P_{rel} \rightarrow C_\mu$ ) :
  - impingement of the curved jet moves upstream from upper shear layer towards flap
  - negative vertical velocity above the flap

*Vertical mean velocity within the back-flow*



- **Side shear layers are free from disturbances.**
- **Upper shear layer is linked to the curve jet**
- **Interaction curved jet/upper shear layer play a role in the development of turbulence?**

# Medium and high frequency forcing : turbulence in the shear layer

*without control*

- ❑ turbulent stresses higher in the upper SL
- ❑ low turbulent activity curved jet ( $|U| \sim 0.3 U_{inf}$ )

*with control*

- ❑ upper SL: turbulence level fewly increased
- ❑ side SL : reduction of turbulent stresses ( $x^* \sim 0.1$ )

Similar conclusions for higher frequency actuation

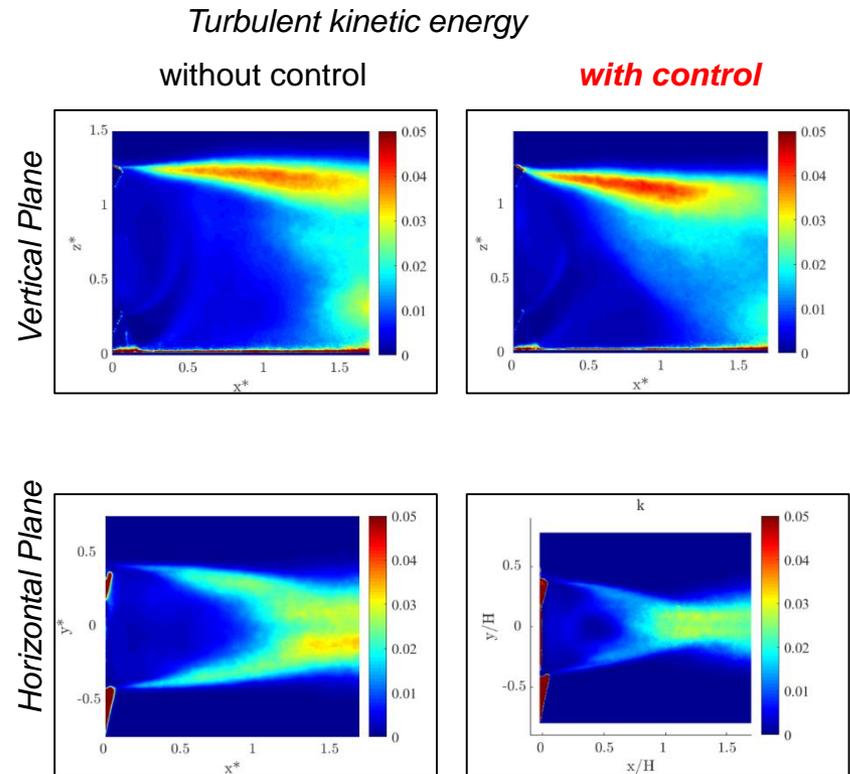
## Summary

- turbulent activity remains low
- Triggering instabilities by the impinging jet ?
- **Effect of forcing on turbulence level related with pressure recovery?**



Vukasinovic et al. JFM 2010

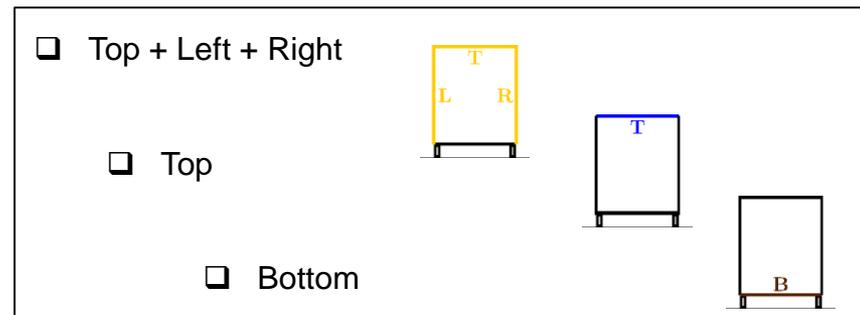
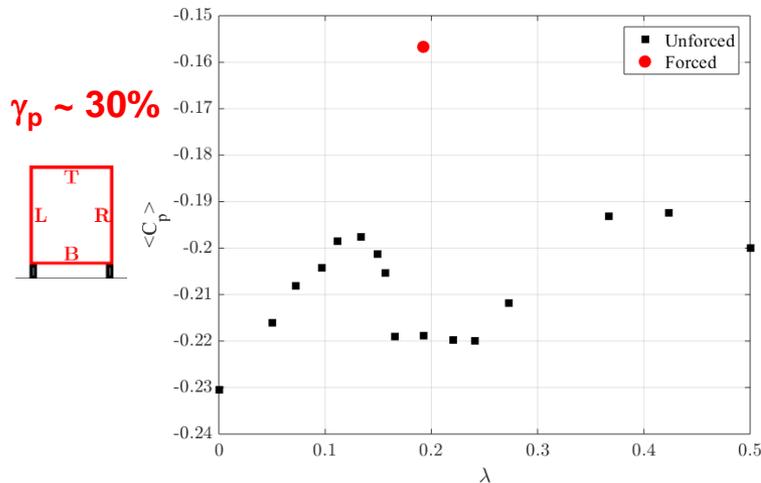
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# Forcing strategy : energetic considerations

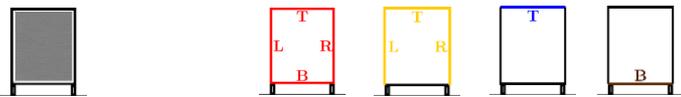
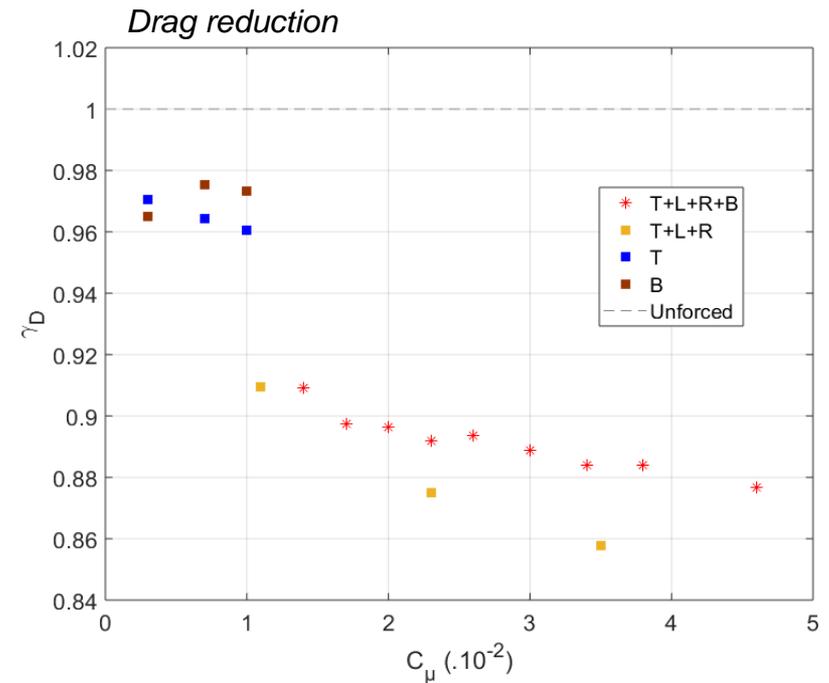
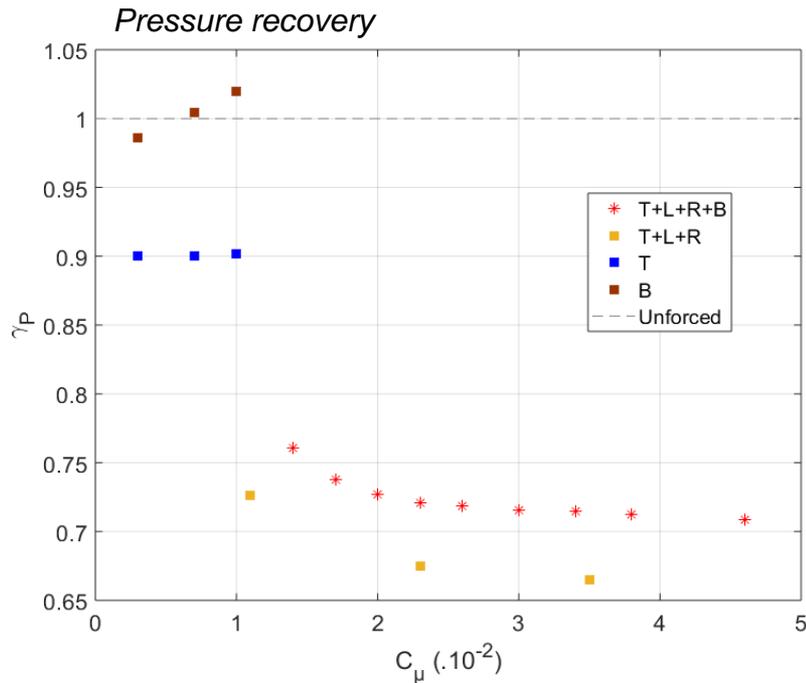
## Global efficiency

- ❑ Ratio between power recovery by drag reduction and power consumption for air compression
  - Acoustic feature (📖 *Michard et al. GDR 2017 Orléans*)
  - Mass flow nearly proportional to:
    - $P_{rel}$  (valve inlet pressure at a fixed frequency)
    - DC (duty cycle)
    - N (number of actuators)
- ❑ Present investigation focuses on reducing number of actuators with fixed values of  $f_{act}$  and DC
- ❑ Local forcing ? → limited number of actuator rows



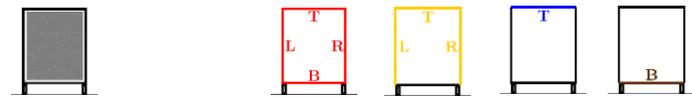
# Forcing strategy : energetic considerations

Evolution of pressure recovery and drag reduction for different control strategies (& increase control intensity)



$$\gamma_P = \frac{\overline{C_{Pb}}}{\overline{C_{Pb0}}}$$

28%	33%	10%	0%
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$$\gamma_D = C_x / C_{x0}$$

12%	12%	3%	2%
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Results matching with previous analysis of the **forcing effect on the individual shear layers**

# CONCLUSIONS and PERSPECTIVES

- Analysis of class III wake topology → Importance of the curved jet
- Global forcing flow results:
  - Wake dimensions reduction
  - No vertical symmetrisation (natural asymmetric flow) since injection momentum is not modifying the underbody flux
  - Angle deflection more important in the side shear layers
  - Low level of turbulence compared with a class IV case
  - Low effect of actuation on the turbulence level
- Control strategy :
  - TLR more performant for base pressure recovery and drag reduction
  - Similar results for class IV case but caused by similar mechanism ?

## Perspectives

Better understanding of:

- unsteadiness
- relation between control parameters and pressure distribution around the flap (and deflection angle of the flow near the flap)
- performance of the control strategies for other classes (e.g. topology in class I)



# THANKS

*GDR Contrôle des décollements- IMFT, Toulouse – 8-9<sup>th</sup> November 2018*

